One optimal control problem for an unmanned aerial vehicle

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The papers [1, 2] deal with one problem of minimizing the time for a kinematic model of unmanned aerial vehicle moving at a constant altitude. From a kinematic point of view, an UAV flying at a constant altitude is determined by standard Dubins equations [3]. Under additional speed constraints, the flight model of a drone is described by the following system of differential equations:

$$\dot{x} = \cos\theta, \quad \dot{y} = \sin\theta, \quad \dot{\theta} = u,$$
 (1)

with $(x, y, \theta) \in \mathbb{R}^2 \times \mathbb{S}^1$ (where $(x, y) \in \mathbb{R}^2$ is UAV coordinates in the plane of constant height, θ is the angle of deviation from the course) and the control $u \in [-1, 1]$. In [1, 2] this (and more general) time-optimal problem was considered with the following final conditions: the UAV steers to the circle of radius 1 centered at the origin and then moves along it clockwise. Due to such final conditions, choosing a new basis $(\tilde{x}, \tilde{y}, \theta)$ one can simplify the system and obtain the two-dimensional time-optimal control problem:

$$\begin{cases} \dot{\widetilde{x}} = 1 + u \cdot \widetilde{y} \\ \dot{\widetilde{y}} = -u \cdot \widetilde{x} \end{cases}$$
(2)

$$|u| \le 1, \quad \widetilde{x}(t_0) = \widetilde{x}_0, \quad \widetilde{y}(t_0) = \widetilde{y}_0, \quad \widetilde{x}(t_1) = 0, \quad \widetilde{y}(t_1) = 1.$$
 (3)

The solution of this time-optimal control problem is rather complicated [1].

But it turns out that if the both choice of the direction of motion along the final circle is allowed (this corresponds to the time-optimal control problem (2) with two endpoints (0, -1) and (0, 1)), then the solution of the time-optimal control problem is essentially simplified. In this paper, we describe the optimal synthesis and give examples of motion with various initial conditions.

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